

Amendments To The Claims:

Please amend the claims as shown.

1 – 10 (canceled)

11. (new) An open-cooled component for a gas turbine, comprising:
an outer wall exposed to a hot gas;
a first cavity partly defined by the outer wall and for a first medium;
a plurality of through-openings are arranged in the outer wall and the through-openings open into the first cavity on a first side and into the hot-gas space on a second side; and
a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,
wherein the second cavity is formed by supply passages that are provided in the outer wall and are connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.

12. (new) The component as claimed in claim 11, wherein the outer wall has a multiplicity of through-bores, a multiplicity of supply passages running between the bores, and a multiplicity of further transverse passages linking the supply passages with the through-bores.

13. (new) The component as claimed in claim 11, wherein the outer wall has at least two layers which can be connected to one another.

14. (new) The component as claimed in claim 11, wherein the passages are incorporated between two layers in a layer surface.

15. (new) The component as claimed in claim 11, wherein the first cavity is connected to a first fluid source and the supply passages can be connected to a second fluid source.

16. (new) The component as claimed in claim 15, wherein one of the two fluid sources is an oxidation source and the other fluid source is a fuel source.

17. (new) The component as claimed in claim 11, wherein the component is a wall element of a combustion chamber or a blade of a gas turbine.

18. (new) A combustion chamber for a gas turbine, comprising:
a component designed as a wall element, comprising;

an outer wall exposed to a hot gas;

a first cavity partly defined by the outer wall and for a first medium;

a plurality of through-openings are arranged in the outer wall and the through-openings open into the first cavity on a first side and into the hot-gas space on a second side; and

a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,

wherein the second cavity is formed by supply passages that are provided in the outer wall and are connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.

19. A gas turbine, comprising:

a compressor section;

a turbine section; and

a combustion chamber, comprising;

an outer wall exposed to a hot gas;

a first cavity partly defined by the outer wall and for a first medium;

a plurality of through-openings are arranged in the outer wall and the through-openings open into the first cavity on a first side and into the hot-gas space on a second side; and

a second cavity for admixing a second medium, the second cavity being fluidically connected to the through-openings,

wherein the second cavity is formed by supply passages that are provided in the outer wall and are connected via transverse passages to the through-openings designed as through-bores, so that the two media cannot be mixed until inside the through-bores.